

# ENLIGHTEN

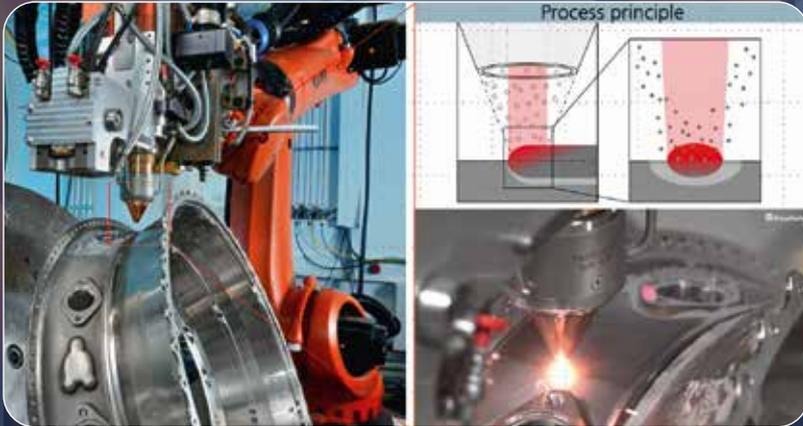
A more efficient and competitive way to space,  
leading to new possibilities in the sky and on earth



## Factsheet #5.1

### Maturation of Additive Manufacturing Methods

# Laser-based Directed Energy Deposition (L-DED)



In laser-based Directed Energy Deposition (L-DED) an additive material feed-stock, either metal powder or wire, is introduced to a melt pool, which is formed by laser radiation. By the relative movement between processing optics and the processing surface, a weld bead is deposited. This process principle allows the versatile application of L-DED in repair, coating and additive manufacturing of metallic components.

The flexibility of L-DED allows the process development towards application adapted specifications. As such, versatile metal alloys, relevant to space and aviation, can be processed. The process can be adapted towards a high productivity production with typical deposition rates of > 2 kg/h or towards the additive manufacturing of filigree structures with wall-thicknesses of < 1 mm. Furthermore, due to the process principle, the technology is scalable between the processing of small components till the manufacturing of large parts like a rocket nozzle.

Especially the additive manufacturing of all nozzle features avoids the current challenges of the production of such a component, which involves multiple processing stations like casting, welding and milling.

### Pros and Cons

- ⊕ Scalable built size, not restricted to smaller components
- ⊕ High rate of material utilization
- ⊕ Energy efficient, laser-based process
- ⊕ Quality control with process sensors can be implemented
- ⊕ Can be automated

---

- ⊖ High degree of process complexity  
Expert knowledge required
- ⊖ Cost intensive metal powder with poor carbon footprint
- ⊖ Long deposition time for big components

### Companies involved



### Key parameters

- 📏 Scalable to bigger built sizes
- 📄 Feasibility of multi material processing
- 🌀 Feasibility of deposition of filigree wall-thicknesses (~ 1 mm)
- 🏭 Typical deposition rate of > 2 kg/h
- 🔧 Resulting material properties at least equal to conventionally produced materials

from

**TRL 3** >>>>

lab scale application for subscale rocket engine elements feasible

to

>>>> **TRL 5**

production of hardware representative of the application and adequate characterisation

### Expectations for testing phase

🕒 **0,6x** lead time for a rocket nozzle

💰 **0,5x** cost when compared to conventional manufacturing processes

🏗️ **higher degree of freedom regarding the dimensional restrictions and choice of materials**